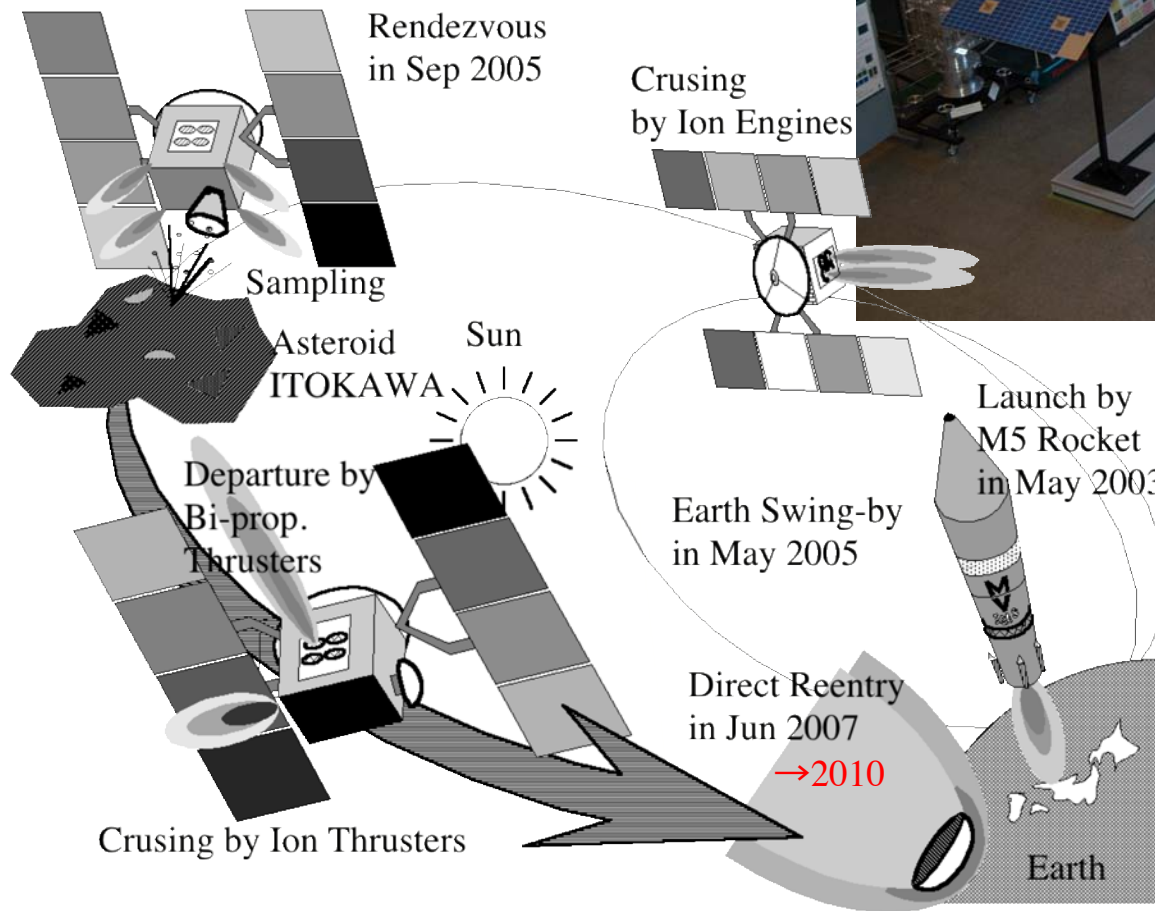
An artistic illustration of the Hayabusa mission. The Hayabusa spacecraft, with its gold-colored body and large blue solar panel arrays, is shown in the foreground, firing its engines. In the background, a large, irregularly shaped grey asteroid floats in space. To the left, a small globe of Earth is visible against the starry backdrop of space.

Hayabusa Asteroid Sample Return Mission

H. Kuninaka
JSPEC/JAXA

Asteroid Explorer “Hayabusa”



Dimensions

: 1.0m x 1.6m x 1.1m

Weight : 380kg(Dry)

Chemical Fuel 70kg

Xe Propellant 60kg

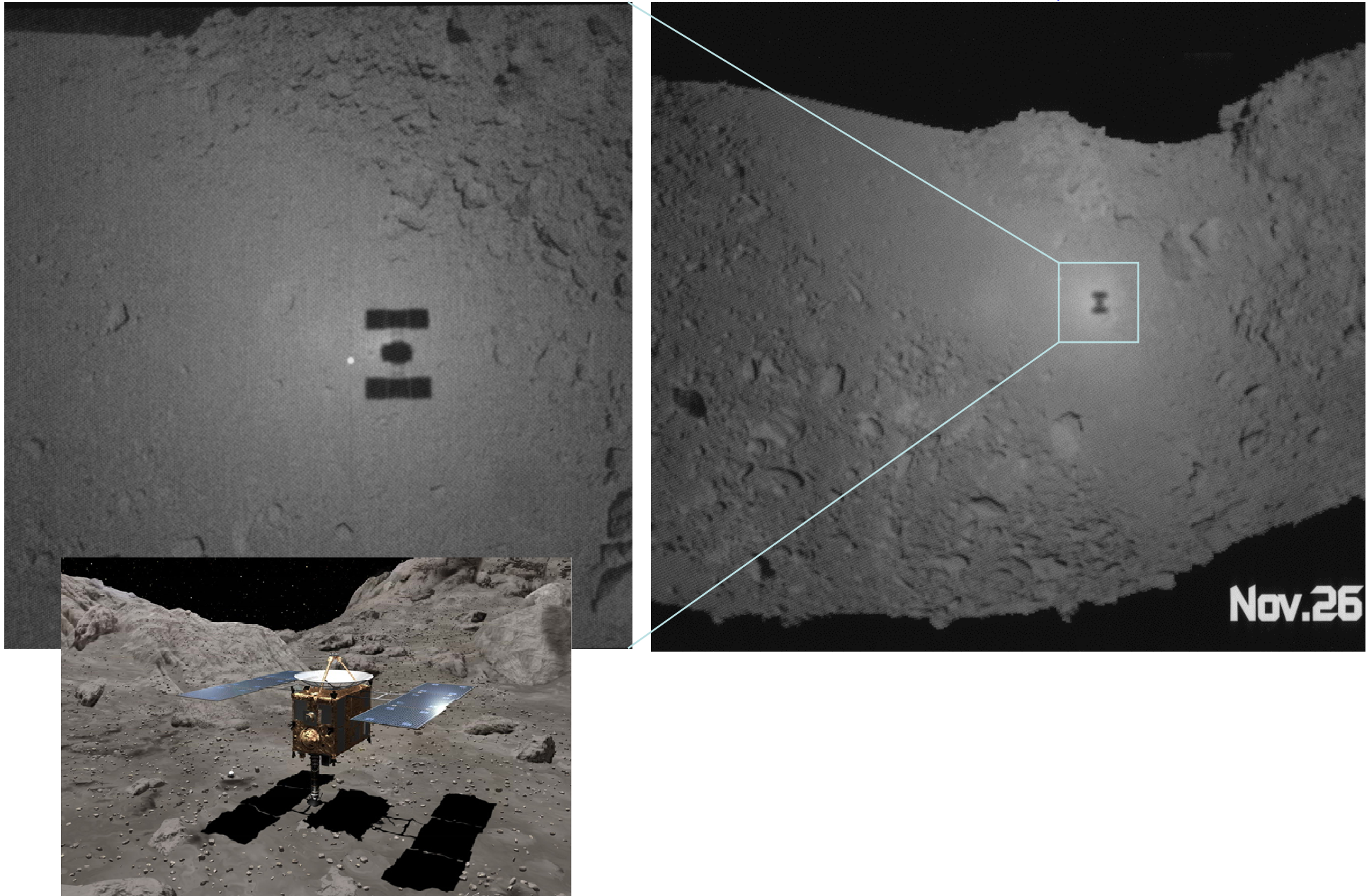
Total 510kg

Electric Power : 2.6kW@Earth

Communication : X band

Closest Snapshot of Asteroid Itokawa

Touch-Down and Lift-off on November 20, 2005



Homeward Journey by Ion Engines from Apr 07

Malfunction Hydrazine Thrusters

Two of three Reaction Wheels

Space Operation

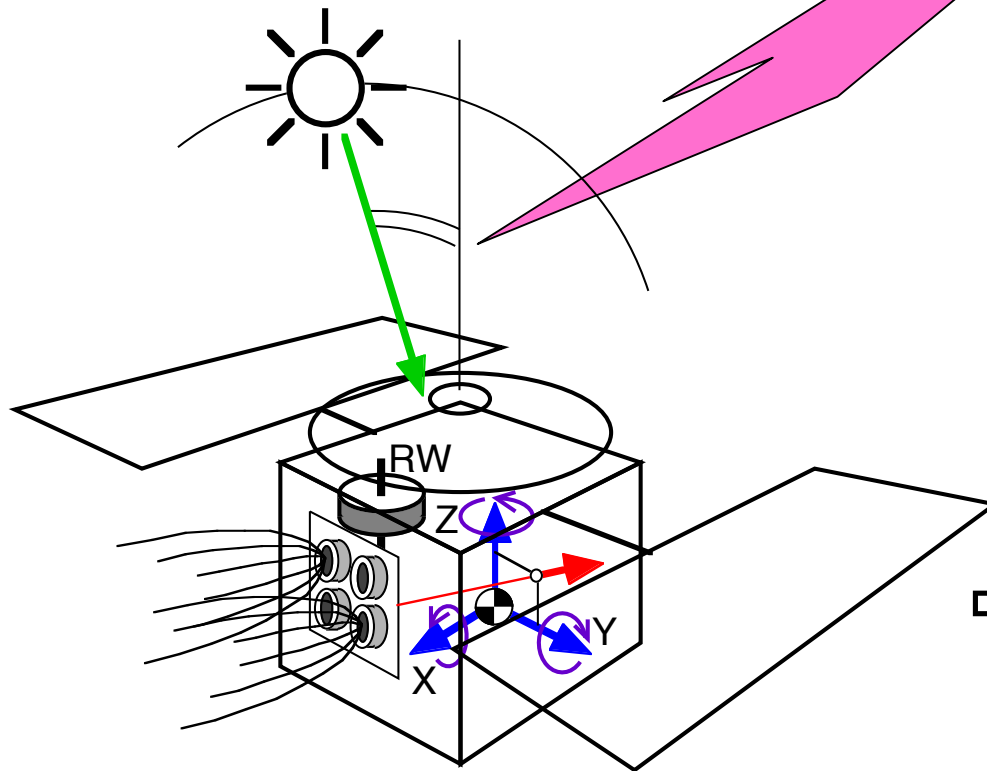
ΔV by Ion Engines

Thrust Vector Control for Y & Z-axis torques

Solar Pressure Torque around X-axis

Single Reaction Wheel for non-spin stabilization

Xenon Cold Gas Jets for Attitude Control

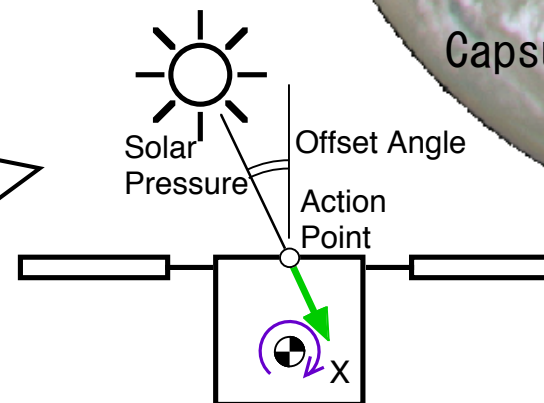


SSOC

Track & Control
Orbit Determine
Orbit Plan



Capsule Recovery

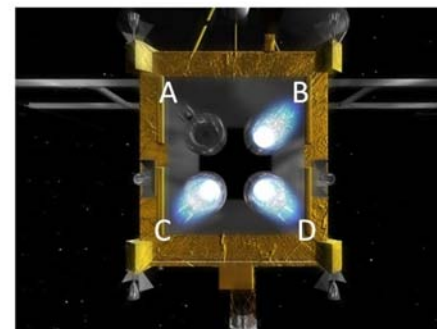
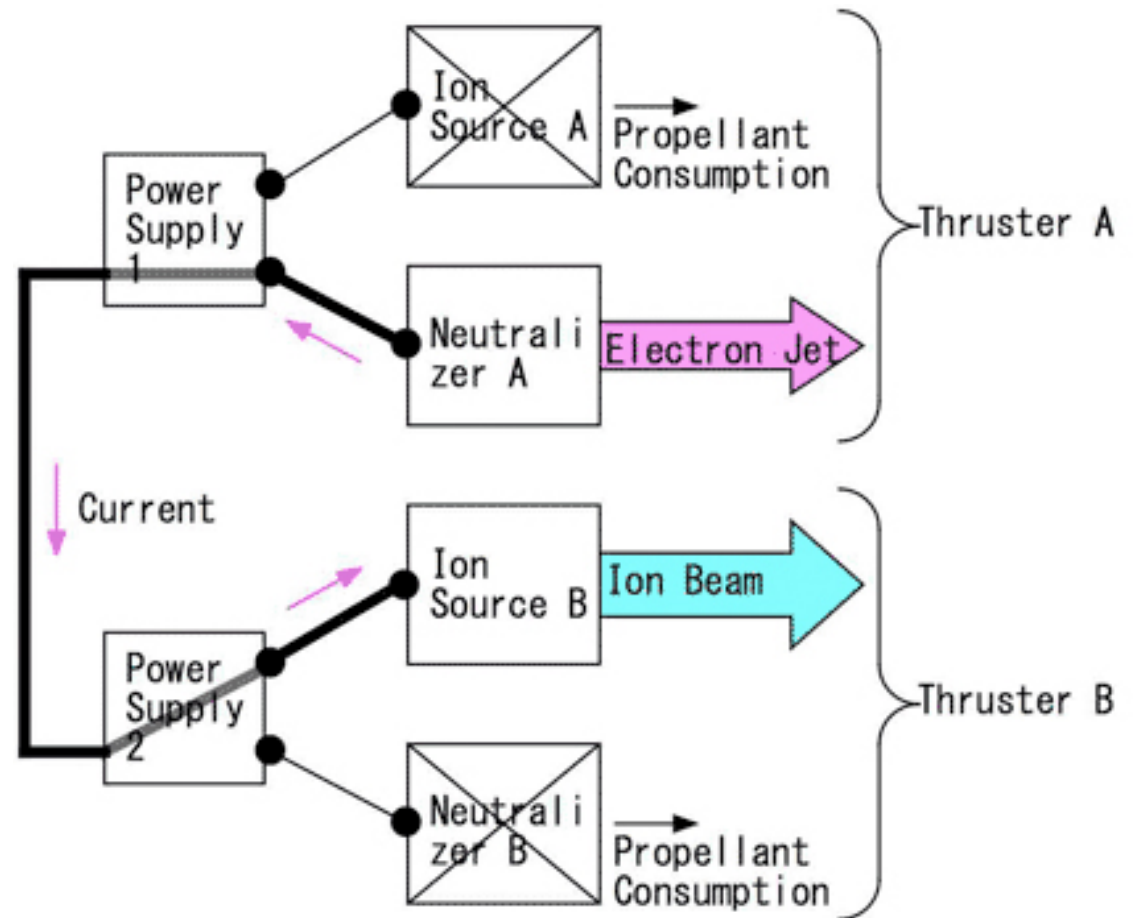


Cross Connected Operation of Ion Engines

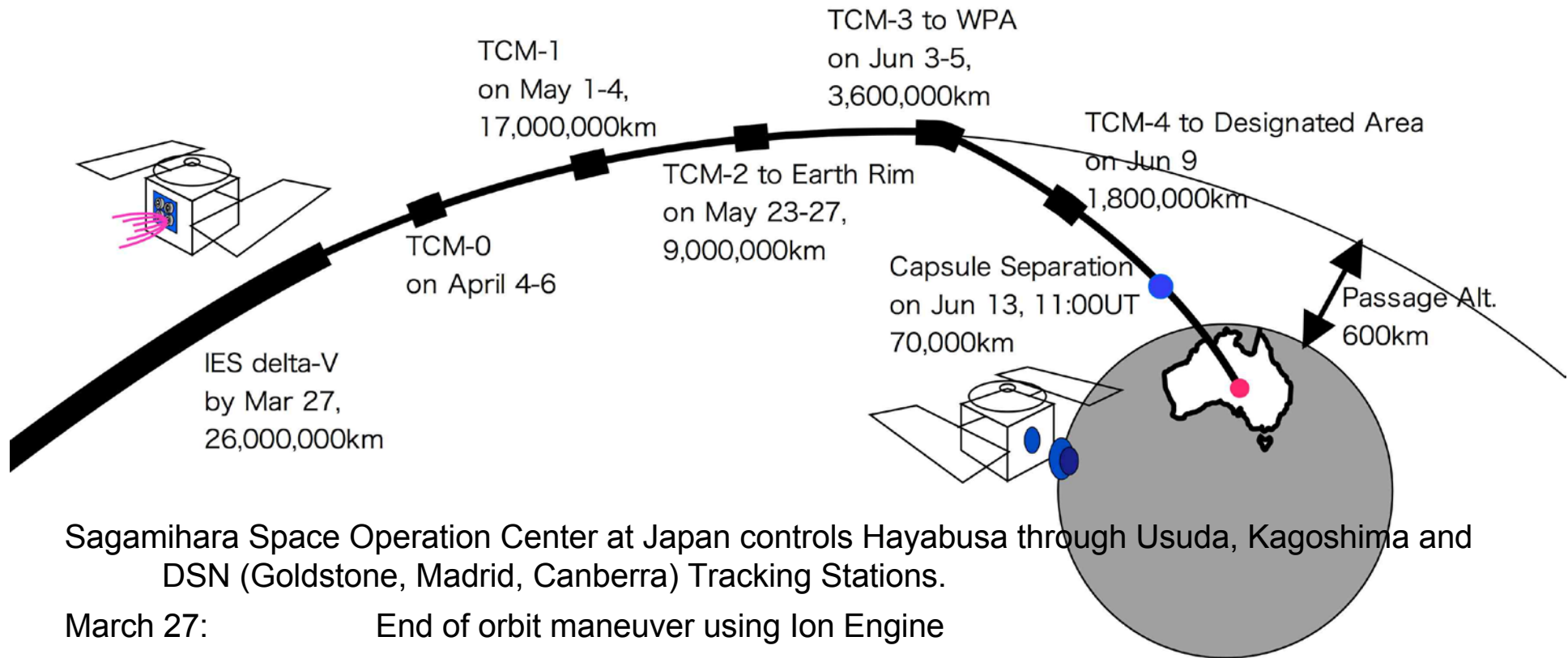
Thruster D was shut off on November 4, 2009.

As counter measure Thruster-B combined with Neut-A generated enough thrust for the rest of the cruise.

The spacecraft was intentionally charged down to negative value, which made Neut-A emit electron current.



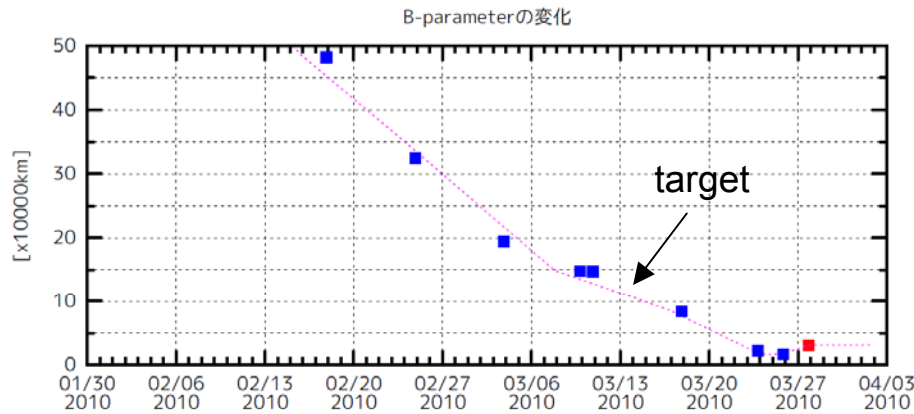
Guidance Scheme to WPA



Sagamihara Space Operation Center at Japan controls Hayabusa through Usuda, Kagoshima and DSN (Goldstone, Madrid, Canberra) Tracking Stations.

- March 27: End of orbit maneuver using Ion Engine
- April 4-6: TCM-0
- May 1-4: TCM-1 guiding to Earth Rim
- May 23-27: TCM-2 precise guiding to Earth Rim above 600km altitude
- June 3-5: TCM-3 changing target from Earth Rim to WPA
- June 9: TCM-4 precise guiding to Designated Area in WPA
- June 13 10:54UT: Separation of Capsule from Hayabusa
- June 13 13:51UT: Reentry of Capsule into atmosphere and Extinction of Hayabusa
(TCM = Trajectory Correction Maneuver)

Hayabusa traversed over south pole and will fly-by at night side of Earth

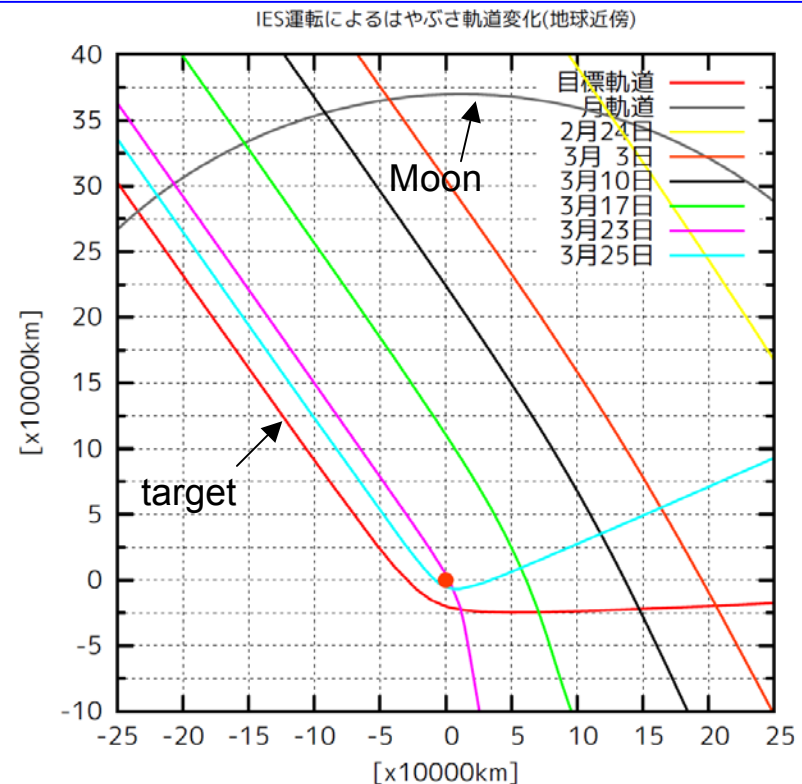
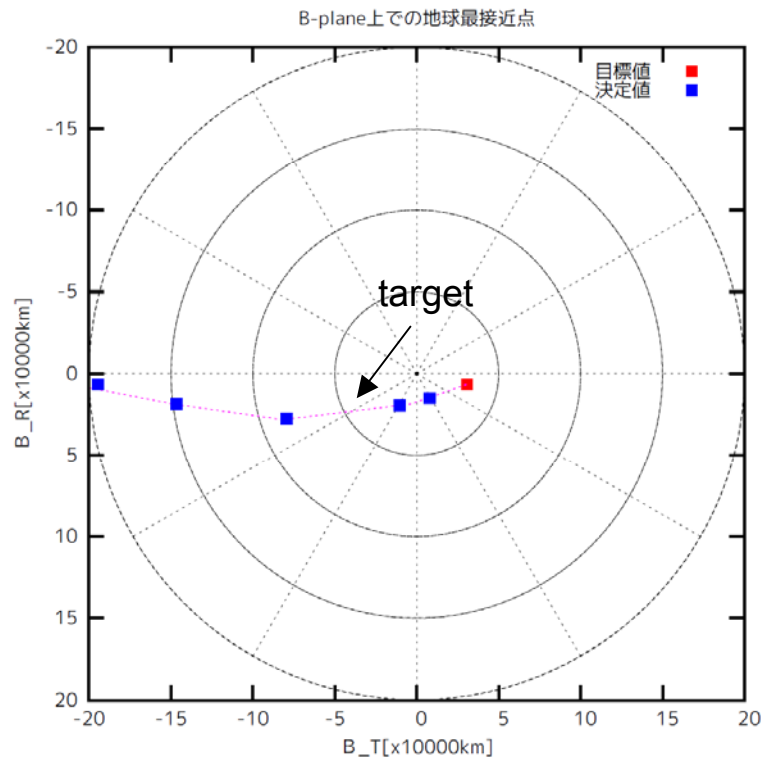


Left: Closest distance to Earth when Hayabusa flies ballistically from the moment designated.

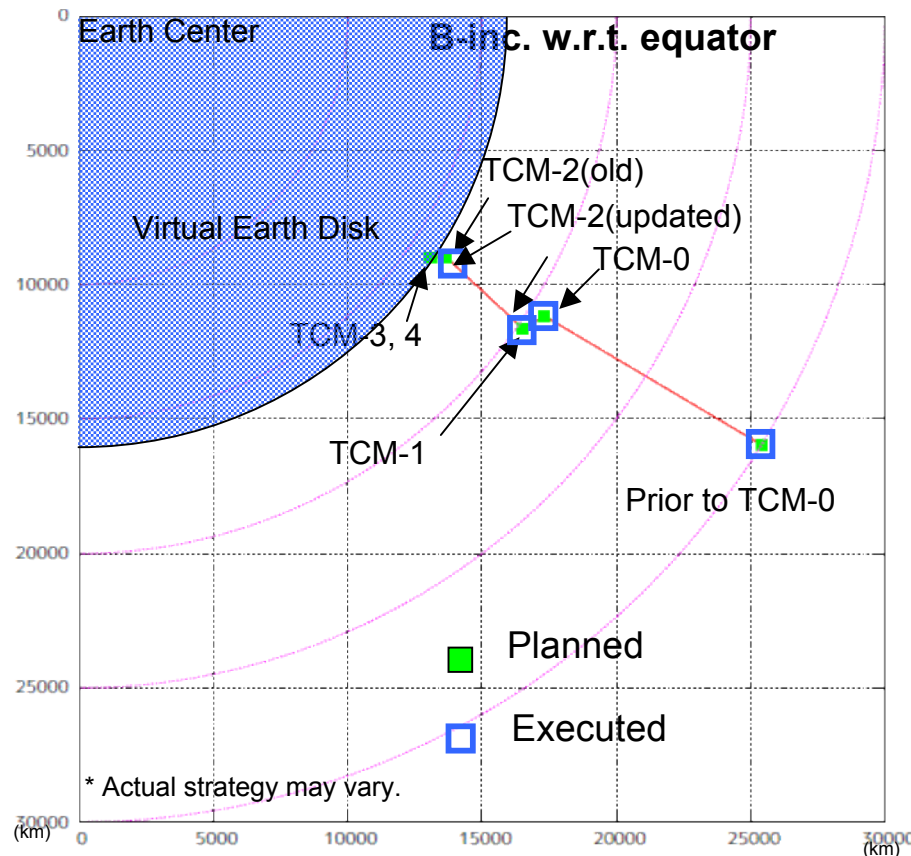
Lower Left: The plane is normal to the approach velocity to Earth, and the horizontal axis corresponds to a crossing line with the equatorial plane. Sun is in the left hand side. Plots show the points where Hayabusa flies through this plane.

Bottom: Trajectories relative to Earth in inertial frame. The plane corresponds to the ecliptic plane.

Notes: Dates designated in the left figure do not indicate the return dates.



Hayabusa completed the correction TCM-2

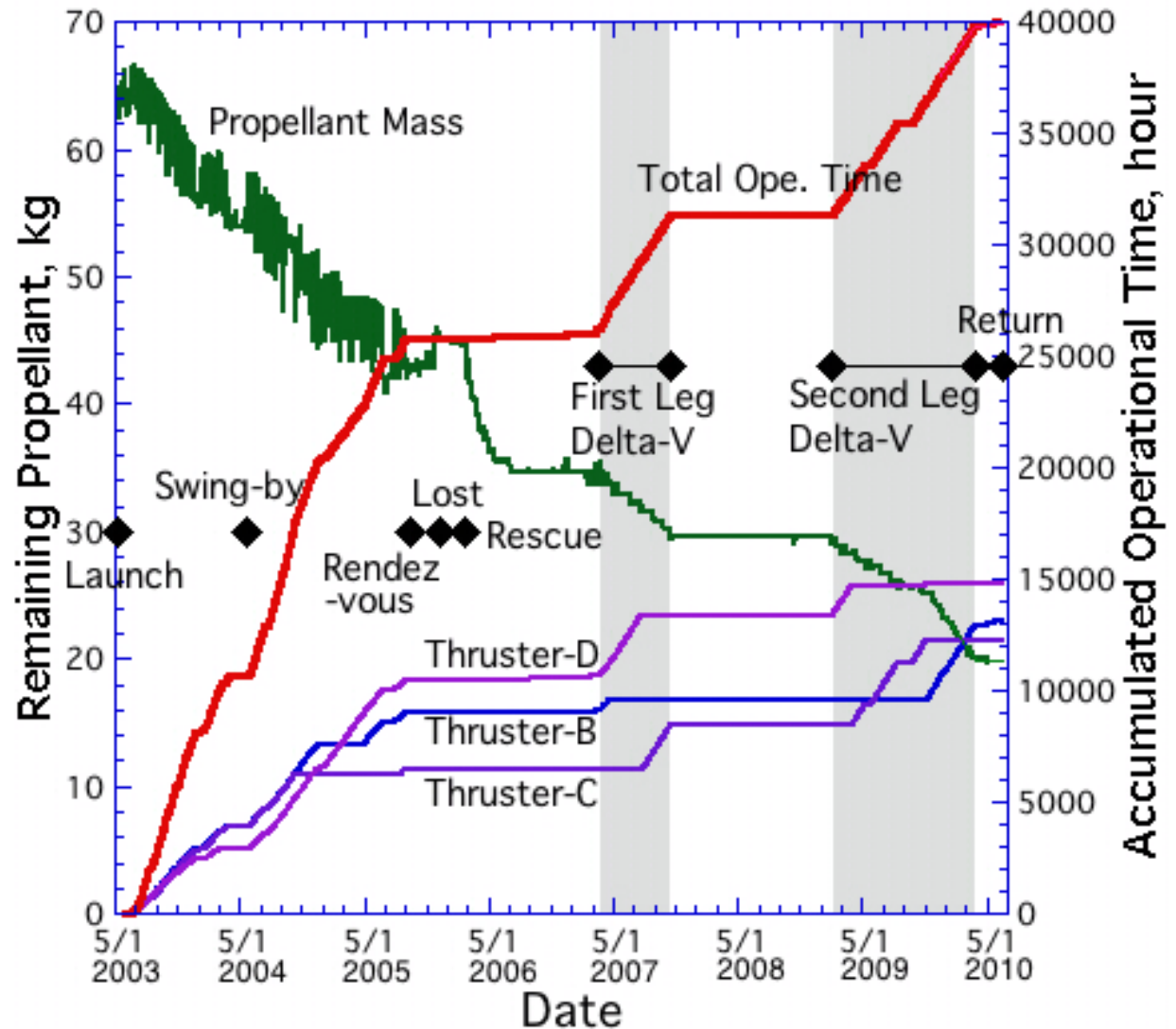


Left : The point where the spacecraft penetrates when no Earth gravity is assumed present. No entry before TCM-3. Virtual Earth Disk is 16 thousand km in radius.

Originally, TCM-2 aimed at the trajectory flying-by at the altitude of 200 km. However, in view of the attitude constraint, the project decided to raise the flying-by altitude to about 630 km. Increase in TCM velocity increment helps the attitude requirement relaxed. And TCM-2 was performed intending the attitude error resides in making the flying-by altitude relatively higher to further relieve the attitude constraint. Currently, the orbit determination is under way.

Summary of Ion Engine Operation

Accumulated Ope. Time:
39,637 hour & unit
Powered Flight Time:
25,800 hour
Longest Operation of a
Single Thruster:
14,830 hour
Power Throttling:
250W-1, 150W
Generated ΔV :
2,200 m/s



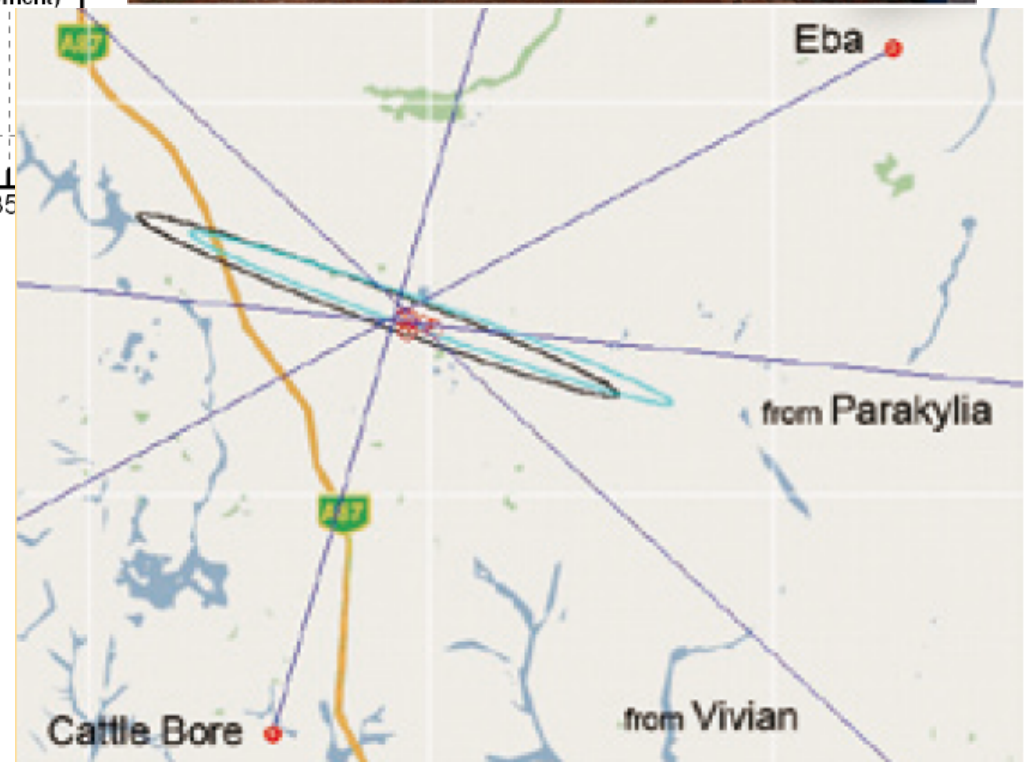
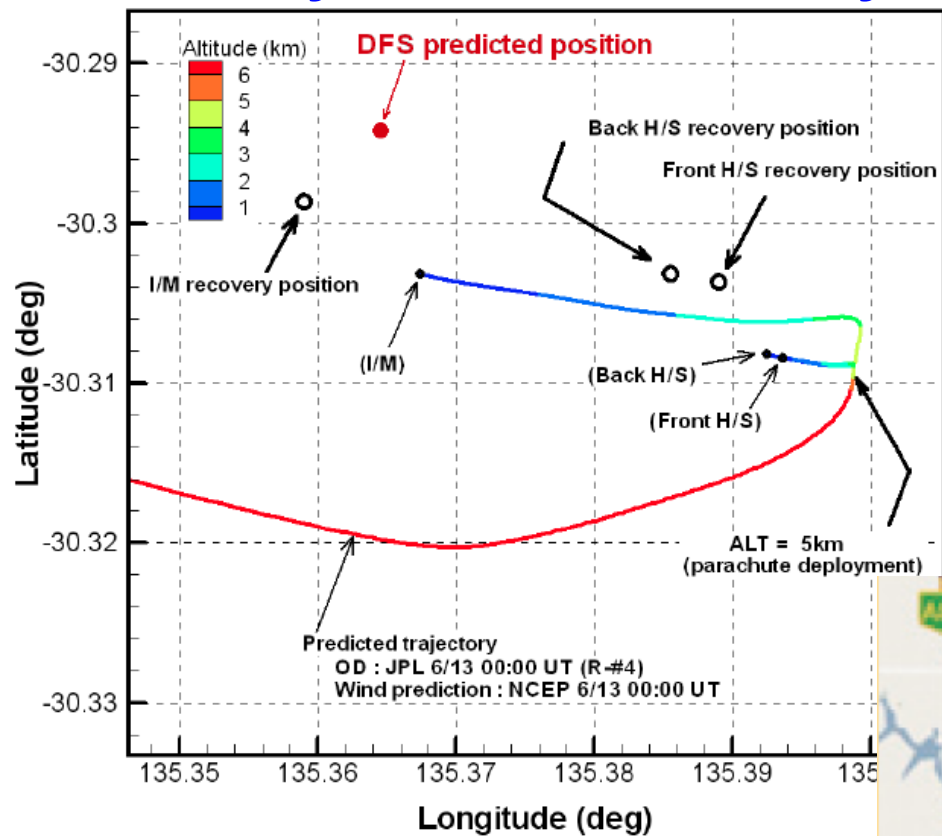
Reentry and Recovery Operation at Woomera

June 13, 14:00 UT

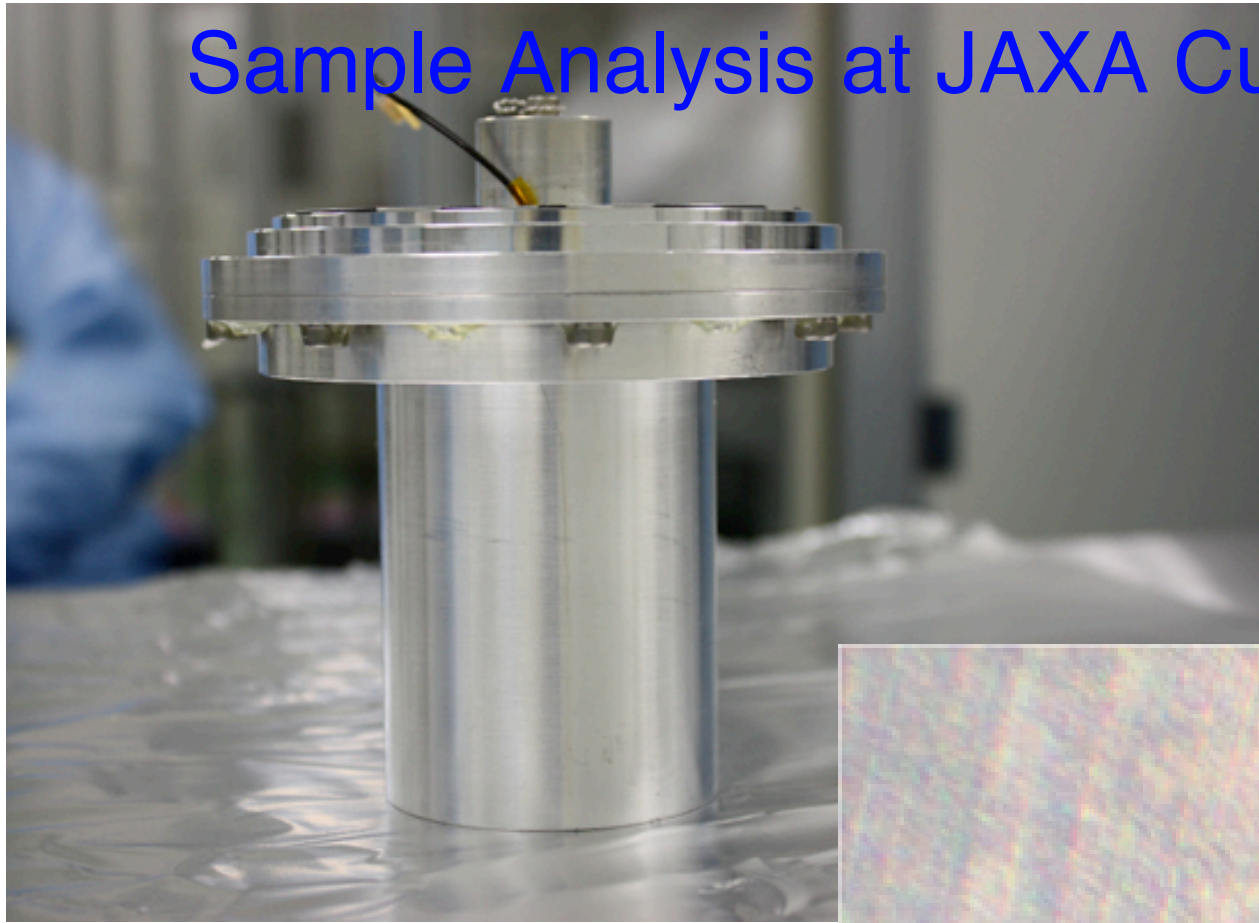


June 14

Reentry and Recovery Operation at Woomera

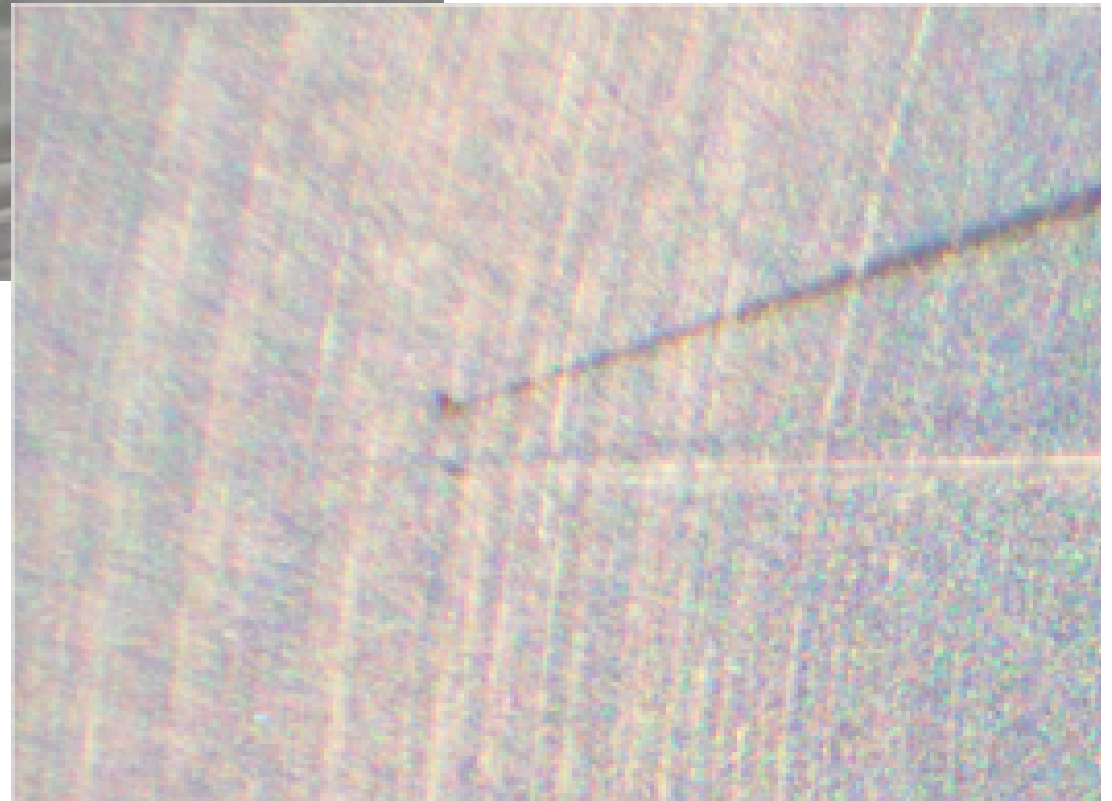


Sample Analysis at JAXA Curation Center

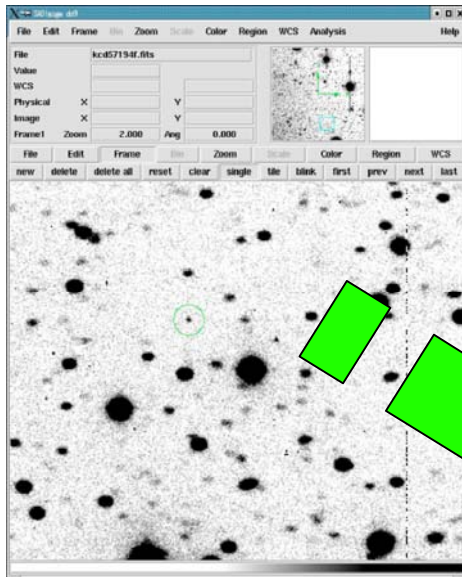


Sample Canister

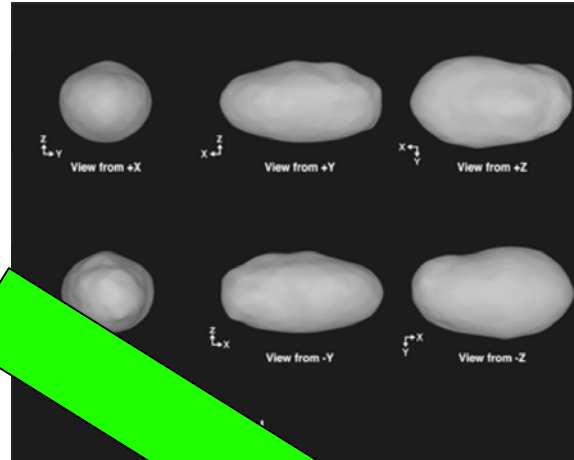
10 μ m particle
inside canister



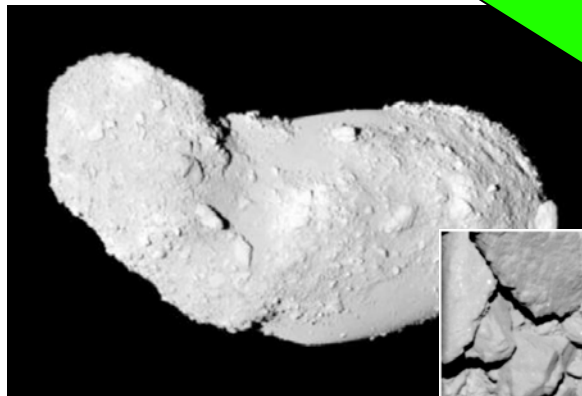
Resolution



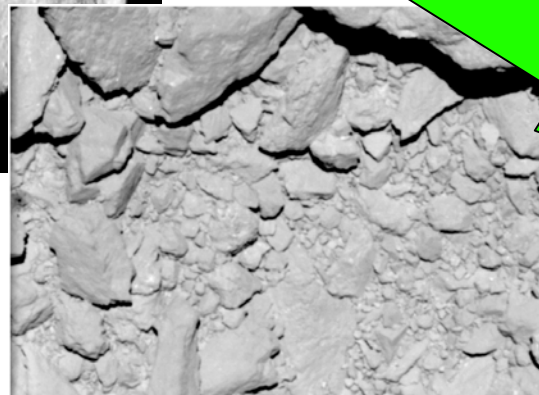
Telescope
AU



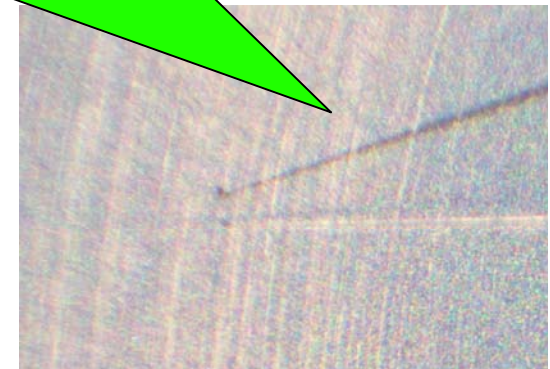
Radar
100m



Rendezvous
1m



Landing, 1mm



Return, 1 μ m

Hayabusa Space Mission was deeply helped by international space communities:

- seamless tracking through NASA Deep Space Network

- OD and EDL analysis by JPL/NASA

- Reentry Airborne Observation by SETI

- COLA analysis by STRATCOM

- kind advice to ground recovery operation by ARES/NASA

- landing authorisation by SLASO/Australia Government

- Woomera operation by AOSG/DoD

- liaison support by BAE Systems Australia

- and so on.

On behalf of Hayabusa Project, I thank you from bottom of my heart for their collaborations.